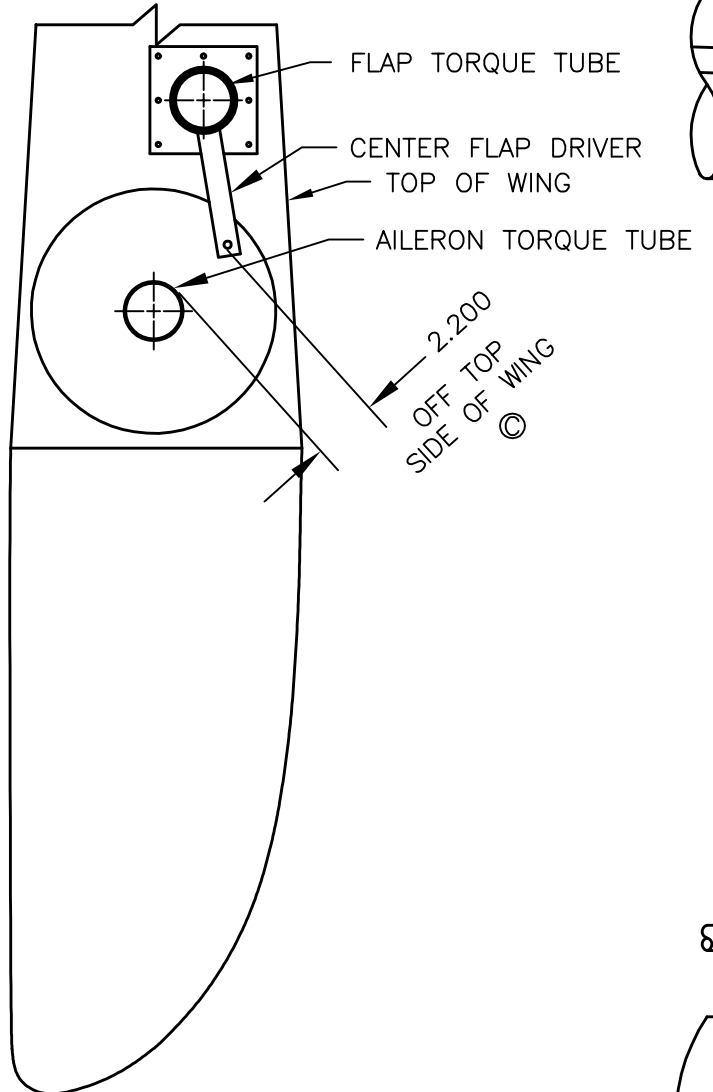


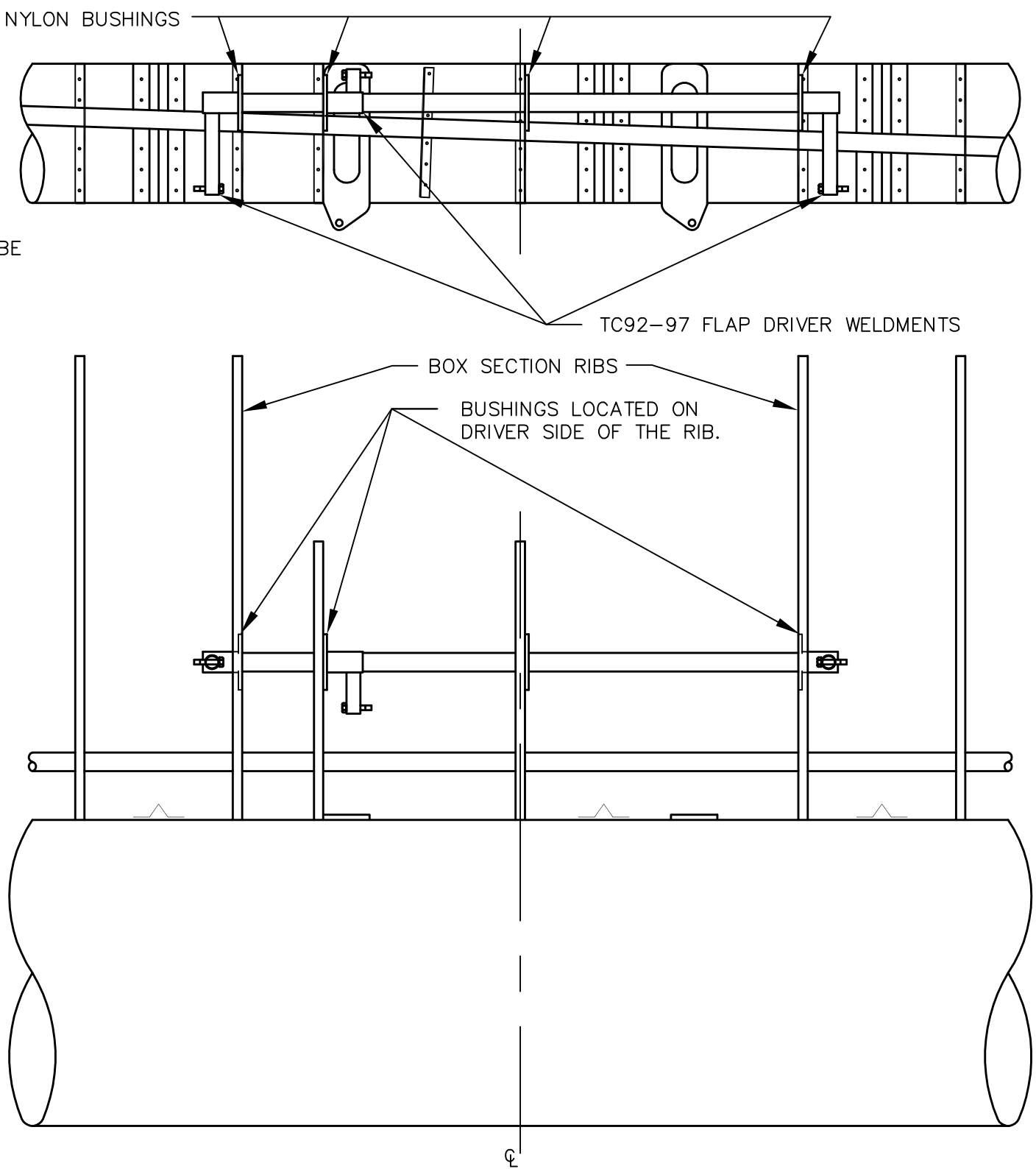
ROTATE FLAP DRIVER AROUND TORQUE TUBE UNTIL THE CENTER OF THE ARM IS 8 1/2 INCHES FROM THE REAR OF SPAR WEB. DRILL (1) HOLES WITH 3/16 DIA. DRILL. USE EXISTING HOLE IN FLAP DRIVER WELDMENT AS A GUIDE. USE AN3-14A BOLT, AN365-1032 NYLON LOCK NUT AND AN960-10 WASHER.

END FLAP DRIVER ORIENTATION



ROTATE FLAP DRIVER AROUND TORQUE TUBE UNTIL BOLT EDGE IS 2.200 INCHES FROM THE AILERON TORQUE TUBE EDGE. DRILL (1) HOLES WITH 3/16 DIA. DRILL. USE EXISTING HOLE IN FLAP DRIVER WELDMENT AS A GUIDE. USE AN3-14A BOLT, AN365-1032 NYLON LOCK NUT AND AN960-10 WASHER.

CENTER FLAP DRIVER ORIENTATION



- NOTES:
- ALL DIMENSIONS ARE INCHES.
 - RIVET BUSHINGS WITH SD43BS RIVETS (1/8 DIA X 21/64 LONG). THE HEAD OF THE RIVETS SHOULD BE AGAINST THE NYLON.
 - DO NOT DRILL MOUNTING HOLES FOR BUSHINGS UNTIL THE WASHOUT OF THE WING IS SET CORRECTLY.

TITAN AIRCRAFT SUPPLY
 1419 STATE ROUTE 45 SOUTH
 AUSTINBURG, OHIO 44010

DRAWN J. PRESTON	DATE 12/11/92	DETAIL NAME FLAP CONTROL SYSTEM ASSEMBLY	SCALE N/A	PART NO.
CHECKED R. INGLE	DATE 3/5/01	ASSEMBLY NAME MANUAL FLAP ASSEMBLY	DRAWING NO.	
APPROVED J. WILLIAMS	DATE 3/5/01	DRAWING NO. B92-INS-0326-C		

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