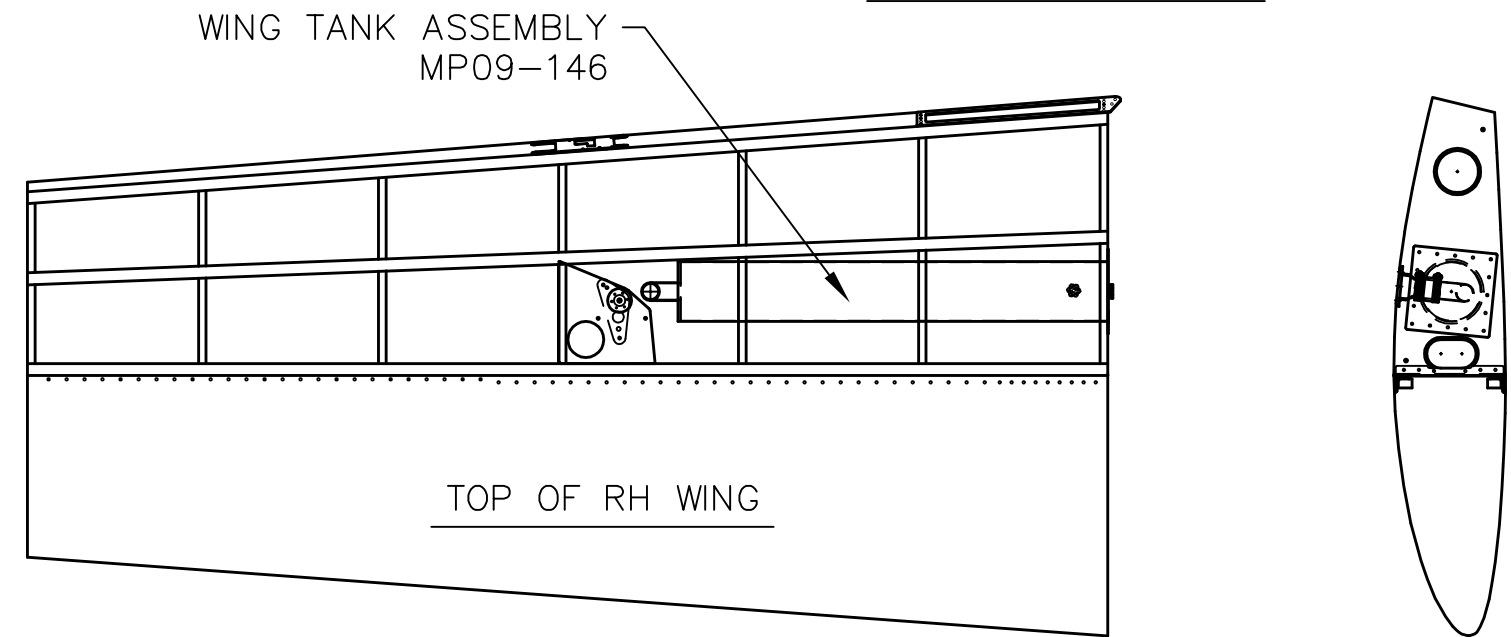
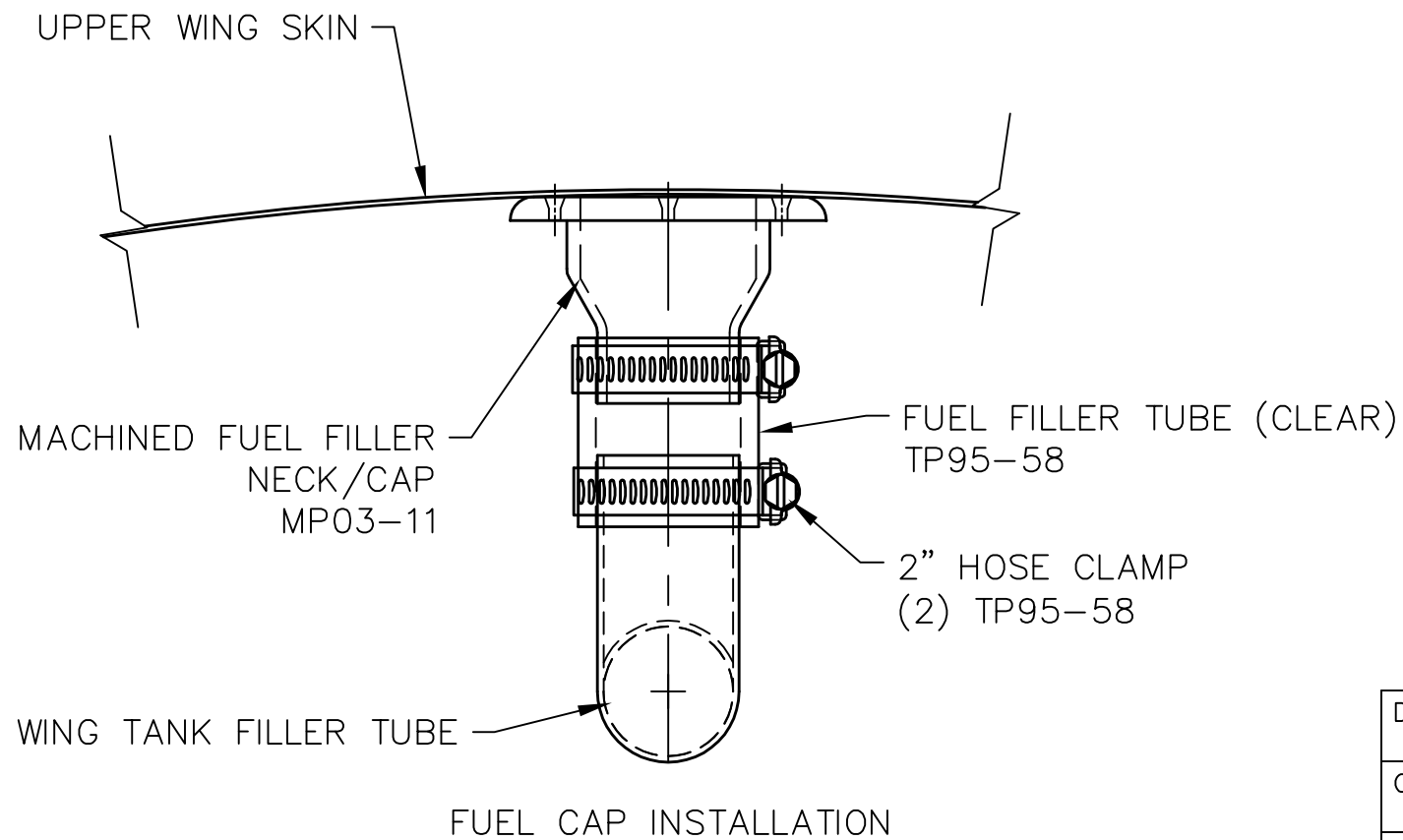
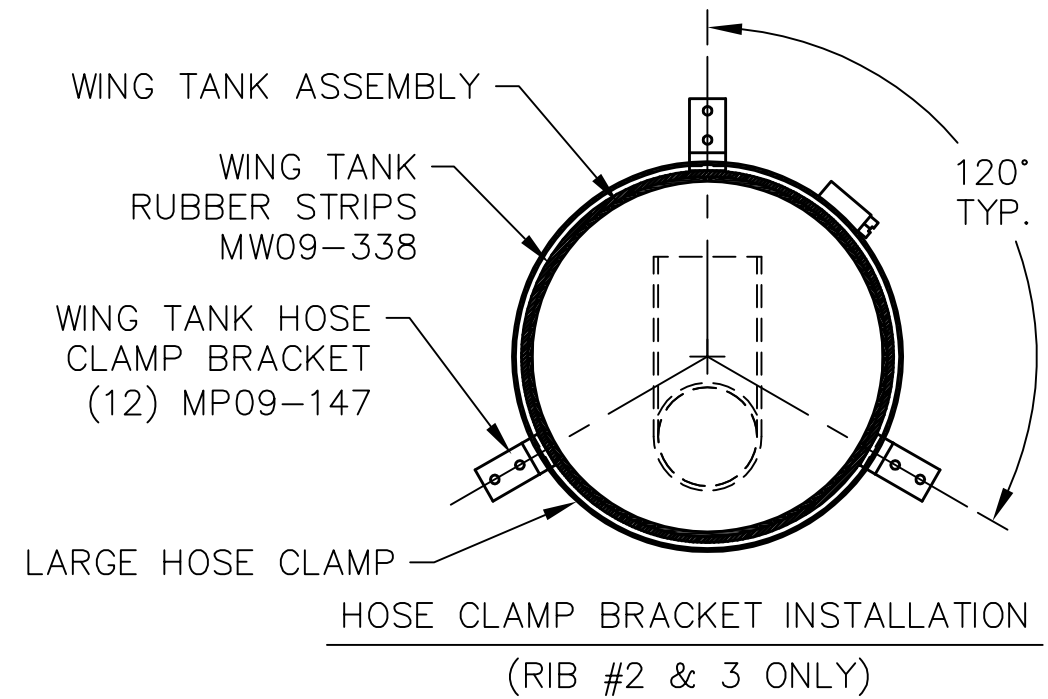


NOTES:

- THE $\phi 5$ " HOLES IN WING RIB'S #1, 2, & 3 MAY NEED TO BE ENLARGED IN ORDER FOR WING TANKS TO FIT PROPERLY.
- USE (MW09-337) TRIM EDGING AROUND ALL HOLES TO PREVENT RIB'S FROM CHAFING THE WING TANKS.
- WITH TANK IN PLACE, POSITION THE WING TANK HOSE CLAMP BRACKETS (MP09-147) AS SHOWN. LEAVING ENOUGH ROOM BETWEEN THE WING TANKS AND MF09-147 FOR RUBBER STRIPS (MW09-338).
- WHEN POSITIONING THE LOCATION OF THE FUEL CAP, TEMPORARILY ASSEMBLE AS SHOWN IN LOWER LEFT, W/OUT HOSE CLAMPS. ROTATE THE WING TANK UNTIL FUEL CAP IS ABOUT FLUSH WITH UPPER SKIN. TRACE FUEL CAP/NECK ON BOTTOM SIDE OF UPPER WING SKIN & DETERMINE THE CENTER OF THE CIRCLE. DRILL HOLE IN UPPER SKIN THE SIZE OF FUEL CAP.
- USE SAME PROCEDURE AS DRAWING T51-04-INS-1089 FOR INSTALLING FUEL CAPS.

REVISIONS				
E.C. NO.	REV.	DESCRIPTION	BY	DATE
	A	DRAWING RELEASED	SH	6/09



TITAN AIRCRAFT SUPPLY
TITAN AIRCRAFT 1419 STATE ROUTE 45 SOUTH
 AUSTINBURG, OHIO 44010

DETAIL NAME	INTERNAL WING TANK INSTALLATION		
SCALE	N/A	PART NO.	
ASSEMBLY NAME	**OPTION**		
PART NO.		DRAWING NO.	W5.5
DRAWING NO.	B T51-09-INS-1841-A		

DRAWN	S. HENDERSON	DATE	6/3/09
CHECKED		DATE	
APPROVED		DATE	

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